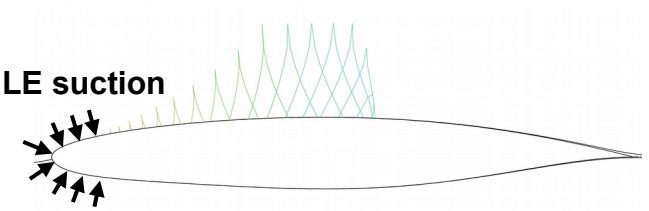
Simulations of HLFC, unswept wing for short range aircraft

Master thesis/Student project

Study objectives:

- Model unswept, finite, HLFC wing for short-range aircraft at flight Mach number 0.4 and Reynolds number 16 Million
- Investigate the effect of suction on BL and pressure distribution
- Investigate the effect of finite wing (aspect ratio) for HLFC application
- Extract BL details and compare to lower fidelity



Tools:

- DLR TAU solver
- Stability tools : Boundary layer (COCO) and LST (LILO) solvers

Required:

- Master student who completed his course work in CFD and aerodynamics

Start date: From Sep. 2021
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