



Kryogen gekühlte Leistungselektronik für Langstreckenflugzeuge Hendrik Schefer

Cryogenically-Cooled Power Electronics for Long-Distance Aircraft, IEEE Access

SPECIAL SECTION ON POWER ELECTRONICS EMERGING TECHNOLOGIES FOR SUSTAINABLE ENERGY CONSERVATION

IEEE Access

Received 4 November 2022, accepted 4 December 2022, date of publication 9 December 2022, date of current version 28 December 2022. Devid Other Medication 30 Devices 30 327 327 328 42

RESEARCH ARTICLE

Cryogenically-Cooled Power Electronics for Long-Distance Aircraft

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This work was supported in part by the Deutsche Forschungsgemeinschaft (German Research Foundation, DFG) through Germany's Excellence Strategy-EXC 2163/1-Sustainable and Energy Efficient Aviation under Grant 390881007, and in part by the Open Access Publication Funds of Technische Universität Brannschweig.

ABSTRACT New aerodynamic aircraft concepts enable the storage of volumetric liquid hydrogen (LH2) Additionally, the low temperatures of LH2 allow technologies such as the superconductivity of electrical components. An increased power density of the onboard wiring harness and the electrical machine can be expected. Nevertheless, the power electronic drive inverter has to deliver high power and high switching frequencies (fPWMs) under challenging conditions. Therefore, knowledge of the electric behaviour of different semiconductor materials under cryogenic temperatures is essential to answer the question: "Are modern power electronics a technology enabler or a system bottleneck?" This publication shows a comprehensive novelty study for cryogenic power electronics based on experimental-driven semiconductor investigations mission profile-based considerations, requirement analyses of superconducting electrical machines, and studies of the cooling concepts. All aspects are discussed within one interdisciplinary publication. A cryogenic system cannot be considered without a feasible cooling concept. Different semiconductor structures based on various materials (silicon (Si), silicon carbide (SiC) and gallium nitride (GaN)) are evaluated for their suitability. The collected data and the literature review draw a technology feasibility studies supported by detailed cooling system analyses and superconducting electrical machine requirements. The power demand and high fPWM lead to a SiC non-cryogenic inverter approach. Due to the detailed cooling system assessment, a loss reduction is achieved by optimising the junction temperature (T1) under various load cases (LCs) out of the mission profile.

INDEX TERMS Long-distance aircraft, fuel cell, liquid hydrogen, cryogenic cooler design, high temperature superconductivity, cryogenic electrical power supply system, cryogenic power electronics, experimental semiconductor comparison, cryogenic inverter feedsign.

ACRONYN	s	CAL
2DEG	two-dimensional electron gas	CFD
AC	alternating current	CO ₂
AlGaAs	aluminium gallium arsenide	D-HEMT
AlSiC	aluminium matrix with SiC particles	DC
ANPC	active neutral point clamped	DCB
BLI	body layer ingestion	DPT
BWB	blended wing body	DUT
		E-HEMT
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approving it f	or publication was Ki-Bum Park.	FEM

 AL.
 controlled axial lifetime.

 Computational fluid dynamics
 carbon dioxide

 Dj.
 carbon dioxide

 VII.
 depletion-mode HEMT

 CC.
 direct tonaled copper

 D/T
 double pubs test

 VDT
 device under test

 VIT
 device under test

 HEMT
 enhancement mode HEMT

 MC
 electromagnetic compatibility

 MF
 finite chement method

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Hendrik Schefer, Wolf-Rüdiger Canders, Jan Hoffmann, Regine Mallwitz und Markus Henke

Dezember 2022 veröffentlicht

30 Seiten, OpenAccess Artikel

Gefördert:



DFG Deutsche Forschungsgemeinschaft

Open-Access-Publikationsfonds

https://doi.org/10.1109/ACCESS.2022.3228161



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3. Halbleiteruntersuchungen [Halbleiter]

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Elektrisches Design

Thermisches Design

Diskussion der Ergebnisse

5. Zusammenfassung [Fazit]





Motivation







Motivation



Herausforderungen:

- Produktion von grünem H₂ und Transport
- Leichte und sichere Speicherung des H₂
- Leistungsdichte der Brennstoffzelle (10 kW/kg)
- Effizienzsteigerung und Kühlung der Brennstoffzellen
- Umrichtergespeiste supraleitende Maschine
- Kryogener oder nichtkryogener Umrichter





Stand der Technik



Analyse

Halbleiter

DC/AC WR

Fazit



Superconducting Component

[1] Modelling of cryogenic cooling system design concepts for superconducting aircraft propulsion





Stand der Technik

Einleitung

Analyse

Halbleiter

DC/AC WR

Fazit



[1] Modelling of cryogenic cooling system design concepts for superconducting aircraft propulsion



[2] Development of High-Power High Switching Frequency Cryogenically Cooled Inverter for Aircraft Applications

[3] MW-Class Cryogenically-Cooled Inverter for Electric-Aircraft Applications



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Wissenschaftlicher Beitrag

Einleitung

Analyse

Halbleiter

DC/AC WR

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SPECIAL SECTION ON POWER ELECTRONICS EMERGING

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RESEARCH ARTICLE

Cryogenically-Cooled Power Electronics for Long-Distance Aircraft

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This work was supported in part by the Deutsche Forschungsgeneinschaft (German Research Foundation, DFG) through Germany's Excellence Strategy-EXC 1263/1-Sustainable and Energy Efficient Aviation under Grant 300881007, and in part by the Open Access Publication Funds of Technische Universitä Brauansdewide;

ABSTRACT New aerodynamic aircraft concepts enable the storage of volumetric liquid hydrogen (LH2) Additionally, the low temperatures of LH2 allow technologies such as the superconductivity of electrical components. An increased power density of the onboard wiring harness and the electrical machine can be expected. Nevertheless, the power electronic drive inverter has to deliver high power and high switching frequencies (fPWMS) under challenging conditions. Therefore, knowledge of the electric behaviour of different semiconductor materials under cryogenic temperatures is essential to answer the question: "Are modern power electronics a technology enabler or a system bottleneck?" This publication shows a comprehensive novelty study for cryogenic power electronics based on experimental-driven semiconductor investigations, mission profile-based considerations, requirement analyses of superconducting electrical machines, and studies of the cooling concepts. All aspects are discussed within one interdisciplinary publication. A cryogenic system cannot be considered without a feasible cooling concept. Different semiconductor structures based on various materials (silicon (Si), silicon carbide (SiC) and gallium nitride (GaN)) are evaluated for their suitability. The collected data and the literature review draw a technology feasibility studies supported by detailed cooling system analyses and superconducting electrical machine requirements. The power demand and high fPWM lead to a SiC non-cryogenic inverter approach. Due to the detailed cooling system assessment, a loss reduction is achieved by optimising the junction temperature (T1) under various load cases (LCs) out of the mission profile.

INDEX TERMS Long-distance aircraft, fuel cell, liquid hydrogen, cryogenic cooler design, high temperature superconductivity, cryogenic electrical power supply system, cryogenic power electronics, experimental semiconductor comparison, cryogenic inverter design.

ACRONYM	IS .	CAL	controlled axial lifetime
2DEG	two-dimensional electron gas	CFD	computational fluid dynamics
AC	alternating current	CO ₂	carbon dioxide
AlGaAs	aluminium gallium arsenide	D-HEMT	depletion-mode HEMT
AlSiC	aluminium matrix with SiC particles	DC	direct current
ANPC	active neutral point clamped	DCB	direct bonded copper
BLI	body layer ingestion	DPT	double pulse test
BWB	blended wing body	DUT	device under test
		E-HEMT	enhancement-mode HEMT
The associapproving it for	iate editor coordinating the review of this manuscript and or publication was Ki-Bum Park.	EMC FEM	electromagnetic compatibility finite element method

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Forschungsfragen:

- Sind Halbleiter der Flaschenhals eines supraleitenden elektrischen Antriebsstrangs?
- Welcher Halbleiter ist geeignet?
- Welche Anforderungen bestehen an die Leistungselektronik?
- Teillastbetrieb
- He-Kühlung und H₂-Kühlung
- Anforderungen der E-Maschine
- Halbleiteruntersuchungen
- Umrichterdesign



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Elektrisches System

Einleitung Analyse Halbleiter DC/AC WR

Fazit



- Hohe
 - Leistungsanforderung
- Teillastbereiche





Elektrisches System

Analyse

Einleitung

Halbleiter

DC/AC WR

Fazit



- Hohe
 Leistungsanforderung
- Teillastbereiche

 $\frac{P_{\mathrm{Mag}}}{P_{\mathrm{Trans,max}}}$ $f = f(f, f_{\text{PWM}})$ $f_{\rm PWM} = 10 \, \rm kHz$ $f_{\rm PWM} = 20 \, \rm kHz$ 0.05 0.05 0.04 0.04 1 0.03 0.03 0.02 0.02 0.01 0.01 15 20 30 40 10 20 $f/kHz \rightarrow$ $f/kHz \rightarrow$

 Magnetisierungs- und Transportverluste sind von dem THD_i abhängig





Thermisches System



Technische Universität Braunschweig

Fazit

Analyse





Thermisches System



ІМАВ

SE²A



Zusammenfassung der Analyse







Zusammenfassung der Analyse









Zusammenfassung der Analyse







Stand der Technik





Stand der Technik



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[24]

Capacitor Banks

Current shunt GaN device

[27]

Cryogenic chamber

Untersuchungsmethodik

Einleitung

Analyse

Halbleiter

DC/AC WR

Fazit

Halbleitermaterialvergleich

Vergleichbare

Leistungsebene

- Untersuchungstemperatur
- Applizierbare Halbleiter: TO 247-3
- Halbleiterparameter:
 - Widerstand im leitenden
 Zustand
 - Sperrspannung
 - Ansprechspannung
 - Schaltverlustenergien





Untersuchungsmethodik

		Device	UDevice	IDevice	Herst.
Einleitung	Halbleitermaterialvergleich		Device	Device	
		Si-	650 V	47 A	A
Analyse	 Untersuchungstemperatur 	MOSFET			
	 Applizierbare Halbleiter: TO 	SiC-	650 V	38 A	A
Halbleiter	247-3	MOSFET			-
	 Halbleiterparameter: 				
DC/AC WR	 Widerstand im leitenden Zustand 	Si-IGBT	650 V	96 A	В
	 Sperrspannung 				
Fazit	AnsprechspannungSchaltverlustenergien	GaN- Cascode	650 V	35 A	С

IMAB

SE²A



Prüfstand und Ergebnisse

Einleitung

Analyse

Halbleiter

DC/AC WR

Fazit



- TO 247-3
- Dynamische und statische Charakterisierung
- -200 °C (flüssigem Stickstoff)





Prüfstand und Ergebnisse

Einleitung

Analyse

Halbleiter

DC/AC WR

Fazit



- TO 247-3
- Dynamische und statische Charakterisierung
- -200 °C (flüssigem Stickstoff)



 Reduzierung der Einschalt- und Ausschaltverluste im IGBT





Prüfstand und Ergebnisse

Einleitung

Analyse

Halbleiter

DC/AC WR

Fazit



- TO 247-3
- Dynamische und statische Charakterisierung
- -200 °C (flüssigem Stickstoff)



- SiC: früheinsetzender Carrier Freezeout
- Optimierung von Si- und GaN-Bauteilen





Zusammenfassung Halbleiter





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Zusammenfassung Halbleiter





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Elektrisches Design Inverter



Analyse

Halbleiter

DC/AC WR

Fazit



- 9 Phasen
 - 2 Level-Topologie
- 3 kV DC / 6,5 kV Bare Die
- 20 kHz
- Thermische Schnittstelle



SE²A



Elektrisches Design Inverter



Analyse

Halbleiter

DC/AC WR

Fazit



- 9 Phasen
- 2 Level-Topologie
- 3 kV DC / 6,5 kV Bare Die
- 20 kHz
- Thermische Schnittstelle









Thermisches Design Inverter





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Thermisches Design Inverter







Zusammenfassung DC/AC WR









Zusammenfassung DC/AC WR







Zusammenfassung DC/AC WR





echnische

Jniversität Braunschweig



Zusammenfassung



Systemanforderung

- Hohe Systemleistungen (Strom und Spannung)
- Geringer THDi \rightarrow Filteraufwand vs. Kein Filter
- H₂: Kühlung vs. He: Kühlung





Zusammenfassung

Einleitung	
Analyse	
Halbleiter	

Systemanforderung

- Hohe Systemleistungen (Strom und Spannung)
- Geringer THDi → Filteraufwand vs. Kein Filter
- H₂: Kühlung vs. He: Kühlung

Halbleiteruntersuchung

- Leitwiderstand und die Verluste der Halbleiter können teilweise durch die niedrigen Temperaturen verbessert werden
- Der Leitwert erreicht nicht die Werte eines Supraleiters

Fazit



Zusammenfassung

Einleitung	•
Analyse	•
Halbleiter	Ha •
DC/AC WR	•
Fazit	•

Systemanforderung

- Hohe Systemleistungen (Strom und Spannung)
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Halbleiteruntersuchung

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- Der Leitwert erreicht nicht die Werte eines Supraleiters

Wechselrichterdesign:

- SiC :Systemleistungen u. Schaltfrequenzen (THD_i) abdecken
- Beide Kühlungsansätze erreichen relativ hohe abgeschätzte gravimetrische Leistungsdichten







Analyse

Halbleiter

DC/AC WR

Fazit

Elektrisches System

 \bullet

Ganzheitlicher Optimierungsansatz → Ströme, Spannungen, Verluste, Fehlertoleranz und Zuverlässigkeit





Einleitung

Analyse

Halbleiter

DC/AC WR

Fazit

Elektrisches System

 Ganzheitlicher Optimierungsansatz → Ströme, Spannungen, Verluste, Fehlertoleranz und Zuverlässigkeit

Halbleiteruntersuchungen:

- Vertikale GaN-Strukturen f
 ür h
 öhere Spannungen
- GaN: Aufbau- u. Verbindungstechnik (integrierte Lösung)



Einleitung

Analyse

Halbleiter

DC/AC WR

Fazit

Elektrisches System

 Ganzheitlicher Optimierungsansatz → Ströme, Spannungen, Verluste, Fehlertoleranz und Zuverlässigkeit

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Wechselrichterdesign:

Thermischen Schnittstellen





Einleitung

Analyse

Halbleiter

DC/AC WR

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Elektrisches System

Ganzheitlicher Optimierungsansatz → Ströme, Spannungen, Verluste, Fehlertoleranz und Zuverlässigkeit

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 ür h
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Wechselrichterdesign:

Thermischen Schnittstellen

Wie geht es weiter am IMAB? Bewilligter Antrag LuFo VI und sea



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Vielen Dank für Ihre Aufmerksamkeit

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